

# Effects of Reaction Mechanisms on Structure and Extinction of Partially Premixed Flames

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The structure and extinction behavior of partially premixed flames in a counterflow configuration are investigated by using five different chemistry models. These include the  $C_1$  and  $C_2$  mechanisms of Peters and Rogg (Peters, N., and Rogg, B., *Reduced Kinetic Mechanisms for Applications in Combustion Systems*, Springer-Verlag, Berlin, 1993, pp. 8–12), a 12-step augmented reduced mechanism, and GRI-2.11 and GRI-3.0 mechanisms. Simulations focus on the comparison of these mechanisms in predicting the structure and extinction of methane–air partially premixed flames over a wide range of strain rates and equivalence ratios, including those corresponding to premixed and diffusion flames. Premixed flame speeds calculated using the  $C_2$  and GRI-2.11 mechanisms are in good agreement with the experimental data, whereas those obtained using the  $C_1$  mechanism show significant differences, especially for fuel-rich conditions. The predicted flammability limits ( $0.5 < \phi < 1.4$ ) are found to be nearly identical for the three mechanisms. In addition, the diffusion flame structures computed using the three mechanisms are essentially the same, except for small differences in the peak temperature values. Results for partially premixed flames indicate that all five mechanisms qualitatively reproduce the double-flame structure associated with these flames. There are, however, notable quantitative differences between the predictions of  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms. For low to moderate strain rates and high levels of air premixing ( $\phi < 2.0$ ), the rich premixed reaction zone for the GRI-2.11 and GRI-3.0 mechanisms is located very close to the fuel nozzle. In addition, the physical separation between the two reaction zones for these mechanisms is significantly larger compared to that for  $C_1$  and  $C_2$  mechanisms. Important quantitative differences are also observed in the predictions of  $C_1$  and  $C_2$  mechanisms. Compared to the  $C_1$  mechanism, the predictions using the  $C_2$  mechanism indicate that 1) the methane consumption and heat release rates in the premixed zone are higher, 2) the flame structure exhibits higher sensitivity to the equivalence ratio, and 3) the two reaction zones merge at a lower equivalence ratio. The extinction strain rates for partially premixed flames are significantly higher using the  $C_2$  and GRI-2.11 mechanisms compared to those using the  $C_1$  mechanism. The effect of radiation heat transfer, computed using an optically thin model on the partially premixed flame structure, is relatively small. Also note that the premixed flame speed plays an important role in determining the stretch rate and, therefore, the structure of partially premixed flames.

## Introduction

**A** PARTIALLY premixed flame is characterized by the presence of synergistically coupled multiple reaction zones. A double partially premixed flame can be obtained by mixing air in less than stoichiometric amounts in the fuel stream and supplying air in the other stream. Compared with the premixed and diffusion (non-premixed) flames, a partially premixed flame is capable of achieving both high-energy supply and low pollutant emissions simultaneously. Therefore, it is important to understand the structure and extinction behavior of partially premixed flames.

Amongst various hydrocarbon fuels, the methane oxidation chemistry has been investigated most extensively, and several detailed mechanisms have been reported. Yang and Puri<sup>1</sup> used a detailed reaction mechanism involving 27 species and 224 elementary steps to examine  $\text{NO}_x$  formation in premixed flames established far from extinction. Williams and Li<sup>2</sup> employed a detailed mechanism containing 177 elementary reactions to investigate  $\text{NO}_x$  formation in partially premixed methane–air flames. Blevins and Gore<sup>3</sup> employed the GRI-2.11 mechanism<sup>4</sup> to investigate the flame structure and  $\text{NO}$  reactions of low-strain-rate partially premixed flames. In their study, a low strain rate was obtained in the large central spatial region of the counterflow by imposing a radial velocity gradient at the boundaries of the computational domain. The GRI-2.11 mechanism considers the chemistry of  $C_1$  and  $C_2$  species, and  $\text{NO}$ , which

involves 279 reactions and 49 species. Peters and Rogg<sup>5</sup> proposed a detailed mechanism consisting of 112 reactions and involving  $C_1$ ,  $C_2$ , and  $C_3$  species. Seshadri and Peters,<sup>6</sup> Seshadri,<sup>7</sup> and Buipham et al.<sup>8</sup> employed reduced mechanisms to analyze the flame structure by using an asymptotic approach. A four-step methane–air mechanism was employed, and the overall flame structure was divided into a preheated zone, a reaction zone, and a postflame zone. Burning velocities and characteristic temperatures were obtained for both the fuel-lean and fuel-rich methane–air mixtures.

In this paper, we investigate the structure and extinction behavior of methane–air partially premixed flames in a counterflow configuration. A schematic of the counterflow flame that contains two reaction zones is shown in Fig. 1. Compared with Bunsen-type and jet flames, a counterflow flame offers many advantages for fundamental investigations. The flame is quite stable and can be treated as a one-dimensional problem. Because the flame can be characterized by a single strain rate, it greatly facilitates the investigation of detailed chemistry models, flow–chemistry interactions, and identification of the dominant reaction pathways. (Note that a commonly used definition of an effective strain rate is the maximum value of the oxidizer-side velocity gradient just before the flame.<sup>9</sup> The same definition is employed in the present study.) The wide and flat flame is also amenable to detailed experimental measurements.

The structure of a partially premixed methane–air flame in a counterflow configuration was first investigated by Yamaoka and Tsuji.<sup>10–12</sup> From detailed measurements, it was established that the flame structure is characterized by two separated reaction zones, a rich premixed zone on the fuel side and a nonpremixed zone on the air side, and that the interactions between these two zones are determined by the combined effect of strain rate  $a_s$  and equivalence ratio  $\phi$ . In addition, it was shown that the rich flammability limit of methane–air mixtures can be extended by using partially premixed combustion, and various stable flame configurations can

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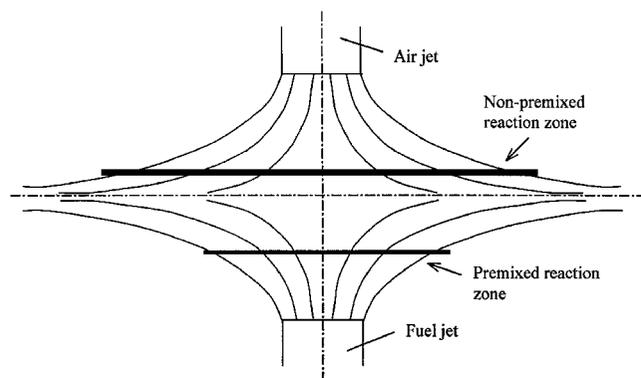


Fig. 1 Schematic of a partially premixed flame in a counterflow configuration.

be achieved by controlling the strain rate and equivalence ratio in the fuel stream. Subsequent studies<sup>13–15</sup> focused on the response of counterflow partially premixed flames to changes in  $a_s$  and  $\phi$ . It was shown that as  $a_s$  is increased at a fixed  $\phi$  (or  $\phi$  is increased at a fixed  $a_s$ ), the two reaction zones tend to merge. A further increase in  $a_s$  at a fixed  $\phi$  then leads to flame extinction, whereas that in  $\phi$  at a fixed  $a_s$  yields a typical diffusion flame structure. Whereas these studies provided fundamental information about the flame structure, the chemical and transport processes involved in the merging of the reaction zones were not investigated in detail. In addition, the role of premixed flame speed in determining the partially premixed flame structure was not examined.

The present study has two major objectives. One is to examine the detailed structure of partially premixed flames and the role of premixed flame speed in determining this structure. Because the flame comprises two spatially separated reaction zones, we also examine the interactions that determine their spatial separation and merging behavior. In this context, we also investigate the response of each reaction zone and, thereby, the flame response to the variations in strain rate and equivalence ratio. The effect of radiative heat transfer on the flame structure and extinction is also characterized.

The second objective is to examine the applicability of several detailed mechanisms for predicting the structure of methane–air partially premixed flames. The mechanisms include the  $C_1$  and  $C_2$  mechanisms of Peters and Rogg,<sup>5</sup> the GRI-Mech 2.11, the 12-step augmented reduced mechanism,<sup>16</sup> and the GRI-Mech 3.0. The  $C_1$  mechanism involves only  $C_1$  species and 52 elementary reactions, whereas the  $C_2$  mechanism considers both  $C_1$  and  $C_2$  species and involves 81 elementary reactions. The 12-step augmented mechanism was reduced from GRI-Mech 1.2 and was validated by Sung et al.<sup>16</sup> for a wide range of combustion phenomena. The GRI-Mech 3.0 is the updated version of GRI-Mech 2.11.

The  $C_1$  and  $C_2$  mechanisms were employed recently by Shu et al.<sup>17,18</sup> to predict the partially premixed flame structure in a planar coflow configuration. Although some quantitative differences between their predictions were noted, the simulations using the two mechanisms were shown to reproduce the measured flame structure over a wide range of conditions. Because the  $C_1$  mechanism has been known to be inadequate for fuel-rich flames, its satisfactory performance in the context of predicting coflow partially premixed flames is surprising and warrants further investigation. Another motivation for using the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms is due to a recent study of Katta and Roquemore,<sup>19</sup> which showed that the structure of coaxial jet diffusion flames computed by using these mechanisms was essentially the same. Because partially premixed flames involves both premixed and nonpremixed combustion, the present study examines the applicability of these mechanisms for premixed, nonpremixed, and partially premixed flames.

### Numerical Model

The counterflow flame configuration employed in the present investigation is shown schematically in Fig. 1. Two coaxial nozzles are placed one above the other. Methane–air mixture flows from the lower nozzle and air from the upper nozzle. The separation distance

between the two nozzles is 2 cm. Both the equivalence ratio and strain rate can be varied independently.

Simulations are performed using the Oppdif<sup>20</sup> and Chemkin<sup>21,22</sup> packages. Oppdif is a FORTRAN program that computes the flame and the flowfield in a counterflow configuration. The two-dimensional axisymmetric flowfield is transformed to a one-dimensional problem by employing a similarity transformation. When the assumptions are used that radial velocity varies linearly with radial distance, and gas properties vary only with axial distance, the dependent variables become functions of the axial direction only. Oppdif solves for the temperature, species mass fractions, axial and radial velocity components, and radial pressure gradient, which is an eigenvalue in the problem. The formulation results in a two-point boundary value problem after boundary conditions are specified at two nozzle exits.

The Chemkin package uses the kinetic mechanism as input and evaluates the chemical reaction rates as well as the thermodynamic and transport properties. The structure of partially premixed flames is analyzed by varying the fuel stream equivalence ratio  $\phi$  and the aerodynamic strain rate  $a_s$ . The effects of  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms on the computed flame structure and extinction behavior are then examined. In addition, a 12-step reduced mechanism and the GRI-3.0 mechanism are examined in the context of predicting the partially premixed flame structure.

## Results and Discussion

### Premixed Flame Speeds Computed with Different Reaction Mechanisms

The flame speed of methane–air mixture as a function of  $\phi$  was computed by using the Chemkin<sup>21,22</sup> premixed flame package. The calculations were performed employing the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms. In Fig. 2, we plot the computed premixed flame speeds  $S_L$  vs  $\phi$  for the three mechanisms and compare them with the experimental data reported by Vagelopoulos et al.<sup>23</sup> The predictions using the  $C_2$  and GRI-2.11 mechanisms show good agreement with the measurements, except for some differences at fuel-lean conditions, whereas those using the  $C_1$  mechanism exhibit significant differences at fuel-rich conditions. Not only does this mechanism underpredict the premixed flame speed, especially for  $\phi > 0.7$ , it also underpredicts the equivalence ratio corresponding to the maximum flame speed. The flammability limits predicted by the three mechanisms are  $0.5 < \phi < 1.4$ , whereas the experimental flammability limits are  $0.5 < \phi < 1.5$ .

### Effect of Premixed Flame Speed on the Partially Premixed Flame Structure

The Oppdif package along with the  $C_2$  mechanism is employed to compute the partially premixed flame structure in a counterflow configuration. The objective is to examine the role of premixed flame speed in determining the partially premixed flame structure. The values of GRAD and CURV, which control the grid size and distribution in Oppdif, are taken as 0.1. The grid independence was

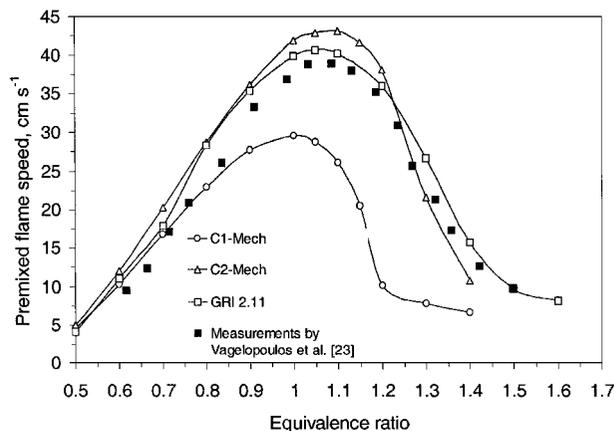


Fig. 2 Comparison of the premixed flame speed calculated by using  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms with the experimental data reported by Vagelopoulos et al.<sup>23</sup>

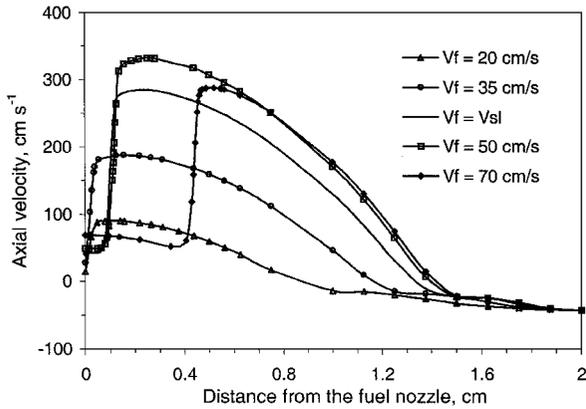


Fig. 3a Axial velocity profiles for different fuel stream velocities obtained by using the  $C_2$  mechanism;  $\phi = 1.1$ , and airstream velocity is  $43.12 \text{ cm/s}$ .

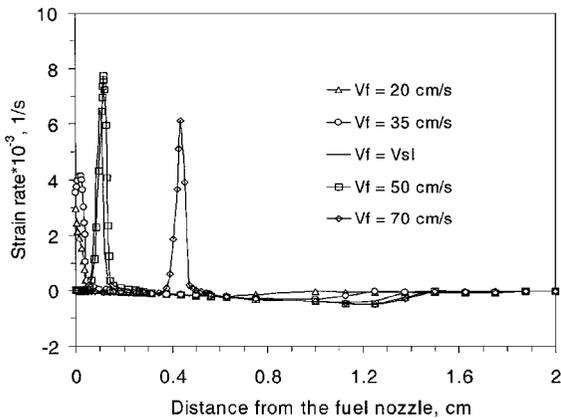


Fig. 3b Strain rate profiles for different fuel stream velocities.

examined by increasing the values of GRAD and CURV by a factor of 5. The difference in the computed flame structure for the two grids was negligible.

Figure 3a shows the axial velocity profiles in the region near the fuel nozzle exit for different fuel stream velocities. The airstream velocity is held constant at  $43.12 \text{ cm/s}$  and  $\phi = 1.1$ , which is within the flammability limit of methane-air mixtures. The flame speed obtained by using the  $C_2$  mechanism for this case is  $S_L = 43.12 \text{ cm/s}$ . When the fuel jet exit velocity  $V_f$  is less than  $S_L$ , the axial velocity increases continuously until it reaches the peak value. For  $V_f = S_L$  and  $50 \text{ cm/s}$ , the axial velocity profile remains flat and then rises sharply as  $x$  approaches the reaction zone. For  $V_f = 70 \text{ cm/s}$ , the axial velocity first decreases to about  $50 \text{ cm/s}$  and then increases abruptly near the flame.

The fuel side axial velocity gradients for various values of  $V_f$  are plotted in Fig. 3b. For  $V_f < S_L$ , the velocity gradient is quite high near the nozzle exit implying that the fuel stream accelerates there. However, for  $V_f > S_L$ , the fuel mixture has to decelerate to a velocity that corresponds to the premixed flame speed. This results in a negative velocity gradient near the nozzle exit. For  $V_f = S_L$ , the velocity gradient is very low. As indicated by Tanoff and Smooke,<sup>13</sup> the maximum fuel-side velocity gradient is an extremely sensitive indicator of the flame character for counterflow partially premixed flames. An important observation here is that the premixed reaction zone of the partially premixed flames has an apparent propagation velocity, which can be approximately determined as the lowest value of the axial velocity just upstream of the premixed reaction zone. Note that the propagation velocity is not the same for all of the mentioned cases because the detailed structure of the partially premixed flame varies with  $V_f$ . Nevertheless, it can be argued that the premixed flame speed plays an important role in determining the detailed structure of a partially premixed flame because it affects the stretch rate. Moreover, because the  $C_2$  mechanism predicts the laminar flame speeds more accurately than the  $C_1$  mechanism, it

is inferred that the partially premixed flame structure is predicted more accurately using the  $C_2$  mechanism than that using the  $C_1$  mechanism.

#### Effect of Reaction Mechanisms on the Partially Premixed Flame Structure

To investigate the effects of reaction mechanisms on the structure of partially premixed flames, simulations are performed for two levels of air premixing,  $\phi = 1.5$  and  $3.0$ , and for a moderate strain rate of  $7.5 \text{ s}^{-1}$ . This strain rate is obtained by setting the exit velocities of the fuel and oxidizer streams at  $10 \text{ cm s}^{-1}$ . The plug flow boundary conditions are employed in the computational model.

Figure 4 shows the computed flame structure in terms of temperature profiles, obtained by using the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms. The corresponding species mole fraction profiles are presented in Fig. 5. These profiles indicate significant differences in the structure of partially premixed flames predicted by the three mechanisms. In contrast, the temperature profiles for the corresponding diffusion flame at the same strain rate indicate an almost identical flame structure for the three mechanisms (cf. Fig. 6). The latter result is consistent with that reported by Katta and Roquemore<sup>19</sup> concerning the effects of different chemistry models on the computed diffusion flame structure. The  $C_1$  mechanism yields a slightly higher temperature compared to that obtained using the  $C_2$  and GRI-2.11 mechanisms, which is due to the endothermic reactions associated with  $C_2$  chemistry that is included in the latter two mechanisms. This result is also in accord with that reported by Katta and Roquemore.<sup>19</sup> An important implication is that the inclusion of  $C_2$  chemistry affects the structure of the rich premixed reaction zone.

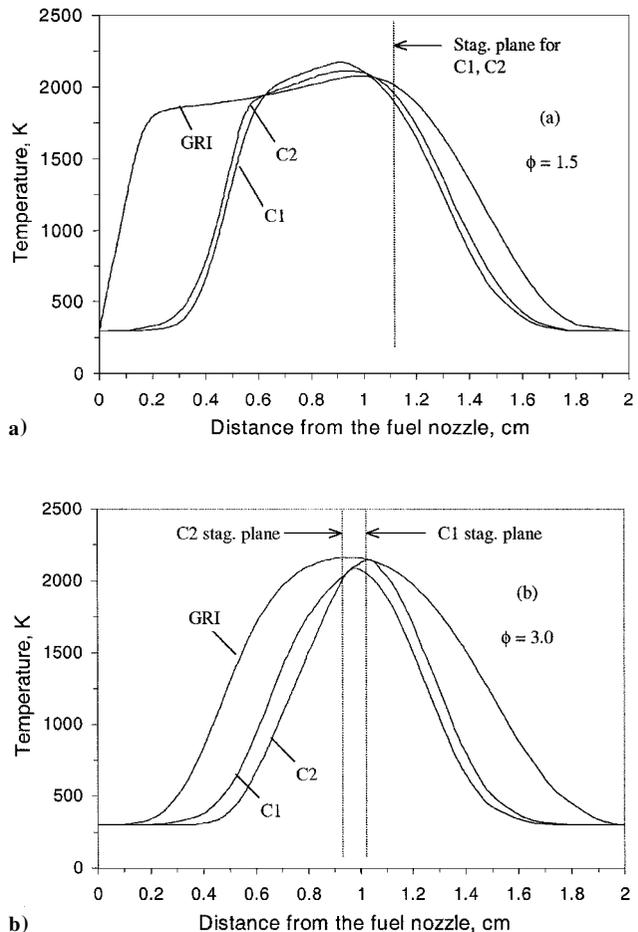


Fig. 4 Partially premixed flame structure in terms of temperature profiles obtained using  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms with fuel and airstream velocities of  $10 \text{ cm/s}$  each and  $7.5 \text{ s}^{-1}$  strain rate for a)  $\phi = 1.5$  and b)  $\phi = 3.0$ .

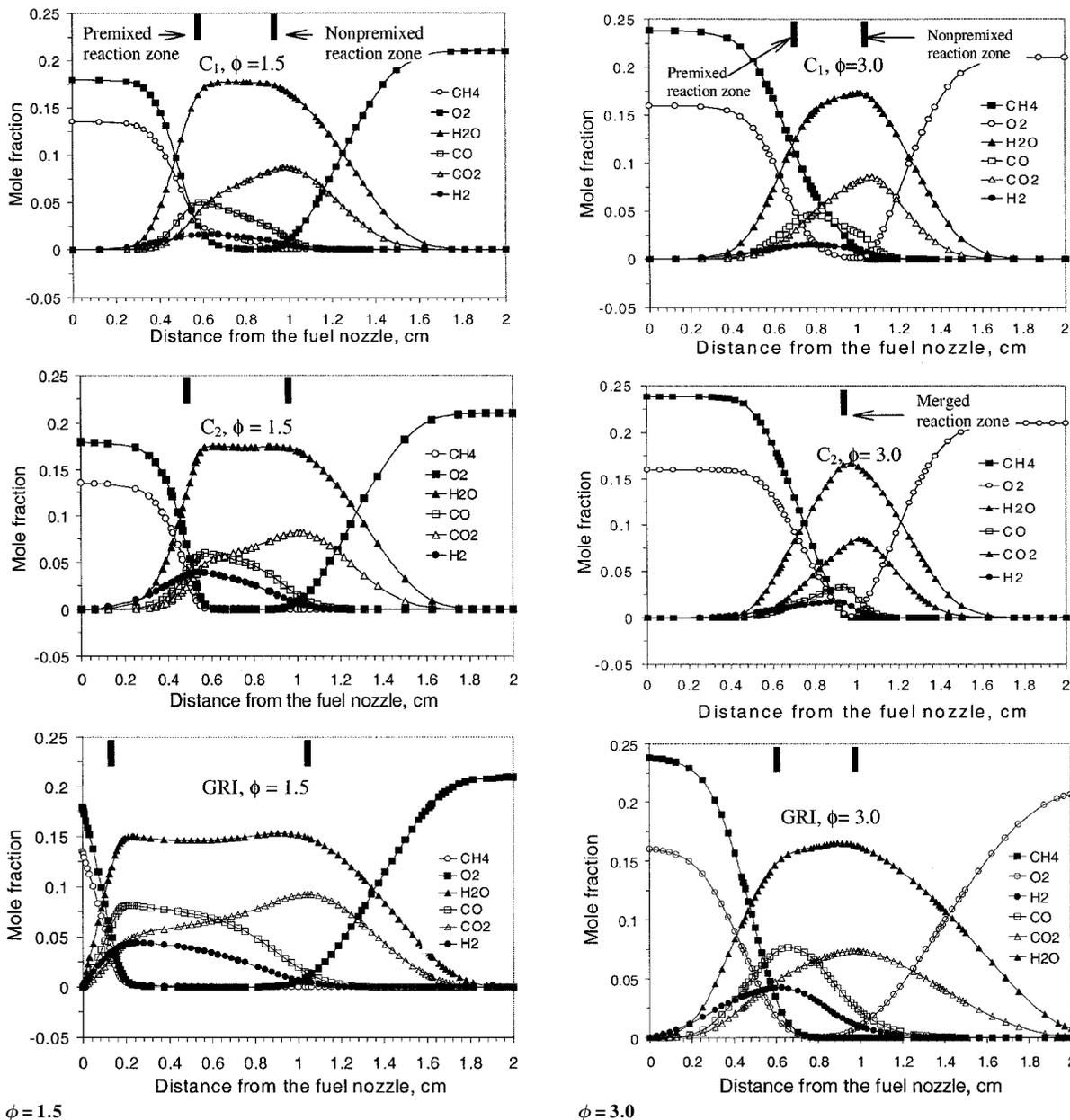


Fig. 5 Major species mole fraction profiles for the partially premixed flame computed using  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms.

The scalar profiles in Figs. 4 and 5 indicate that all three mechanisms reproduce qualitatively the global structure of partially premixed flames obtained in laboratory experiments.<sup>10–12</sup> The global flame structure pertains to the existence of two reaction zones, namely, a rich premixed reaction zone on the fuel side and a nonpremixed reaction zone on the air side, and the synergistic interactions between them. The premixed zone is dominated by the fuel consumption chemistry, whereby  $\text{CH}_4$  and  $\text{O}_2$  react with H and OH radicals to produce CO and  $\text{H}_2$ , which then become the reactant or fuel species for the nonpremixed zone, where they are oxidized to form  $\text{CO}_2$  and  $\text{H}_2\text{O}$ . The nonpremixed zone, being the zone of highest temperature, in turn provides the heat and radical species (such as H and OH) to the premixed zone. The nonpremixed zone chemistry is dominated by the  $\text{H}_2$ - $\text{O}_2$  reactions, CO oxidation, and the production of  $\text{CO}_2$  and  $\text{H}_2\text{O}$ . This is clearly indicated by the peaks in CO and  $\text{H}_2$  mole fractions in the premixed zone and the peak in  $\text{CO}_2$  that occurs in the nonpremixed zone. The peak in  $\text{H}_2\text{O}$  occurs at a lower axial location compared to that in  $\text{CO}_2$  because the  $\text{H}_2\text{O}$  formation rate is higher than that of  $\text{CO}_2$ .

Whereas all three mechanisms reproduce the double-flame structure associated with partially premixed flames, they also exhibit sig-

nificant quantitative differences. In particular, the GRI-2.11 mechanism yields a flame structure that is markedly different from those obtained using the other two mechanisms. At  $\phi = 1.5$ , the rich premixed reaction zone for the GRI-2.11 mechanism is located very close to the fuel nozzle, causing significant heat loss to the nozzle. In addition, the global flame with this mechanism is significantly broader than that with the other two mechanisms (cf. Figs. 4 and 5). This is somewhat unexpected but noteworthy result, especially because the predictions of  $C_2$  and GRI-2.11 mechanisms are in good agreement for both the premixed and diffusion flames.

To examine this aspect, we also computed the partially premixed flame using the GRI-3.0 mechanism as well as a 12-step augmented reduced mechanism.<sup>16</sup> The 12-step mechanism has been reduced from the GRI-1.2 mechanism. Figure 7 presents a comparison of the partially premixed flame structure in terms of temperature profiles computed using the GRI-2.11, GRI-3.0, and 12-step mechanisms. For  $\phi = 1.5$ , the nonpremixed zone obtained with the GRI-3.0 mechanism is located slightly closer to the premixed zone compared to that obtained with the other two mechanisms. The overall differences between the predictions of these three mechanisms are negligible. Thus, an important conclusion from Figs. 4 and 7 is that the partially

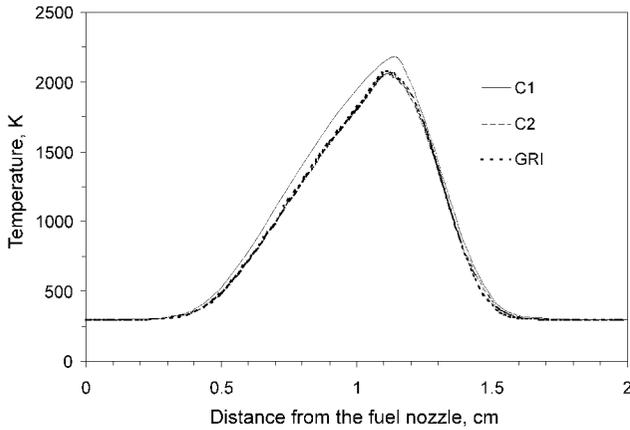


Fig. 6 Temperature profiles for a methane-air diffusion flame computed by using  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms; fuel and air-stream velocities are 10 cm/s each; strain rate is  $7.5 \text{ s}^{-1}$ .

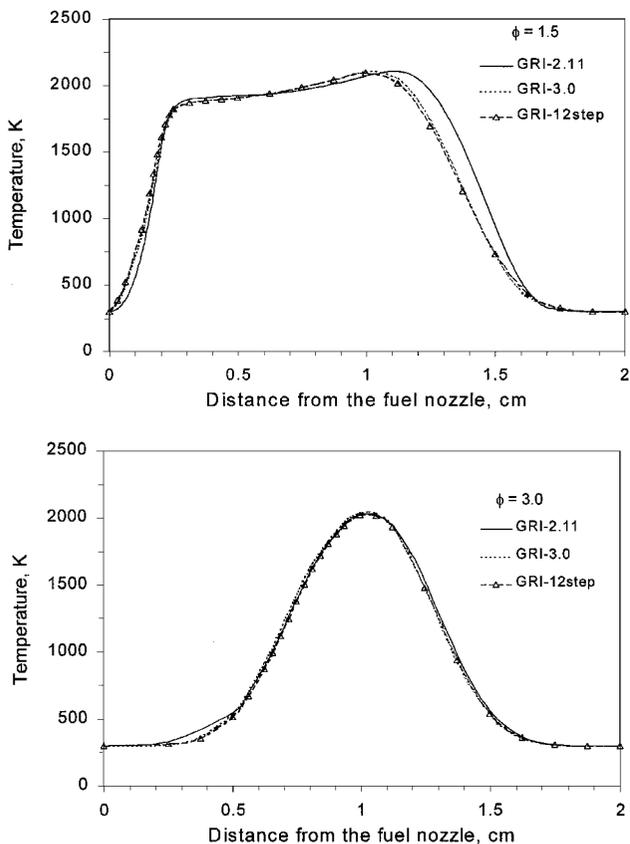


Fig. 7 Partially premixed flame structure in terms of temperature profiles obtained using the GRI-2.11, GRI-3.0, and 12-step reduced mechanisms corresponding to the conditions of Fig. 4.

premixed flame structure obtained using the  $C_1$  and  $C_2$  mechanisms is significantly different from that obtained using the GRI-2.11 and GRI-3.0 mechanisms. Measurements of the counterflow partially premixed flame structure at different levels of air premixing are needed to address these differences.

#### Differences Between $C_1$ and $C_2$ Mechanisms

There are notable differences between the predictions of  $C_1$  and  $C_2$  mechanisms. At higher level of air premixing,  $\phi = 1.5$ , the premixed reaction zone obtained using the  $C_2$  mechanism is closer to the fuel nozzle than that using the  $C_1$  mechanism. This implies that the methane consumption rate in the premixed reaction zone is higher for  $C_2$  mechanism than that for  $C_1$  mechanism. This is

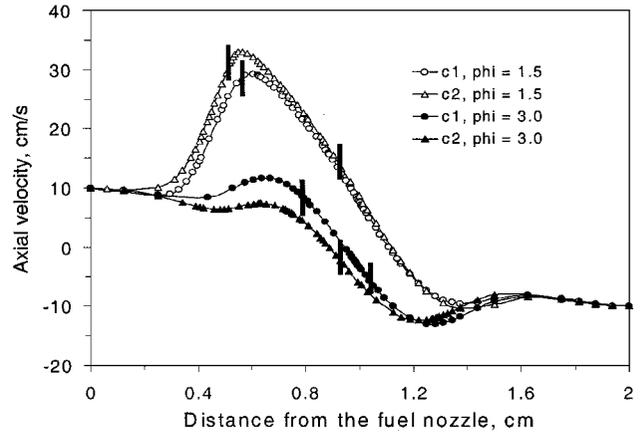


Fig. 8 Axial velocity profiles for the  $C_1$  and  $C_2$  mechanisms corresponding to the conditions of Fig. 4.

confirmed by the  $\text{CH}_4$  consumption rate profiles that are presented later.

At lower level of air premixing, that is,  $\phi = 3.0$ , however, the situation is reversed. As indicated in Fig. 4b, the premixed reaction zone obtained using the  $C_1$  mechanism is located closer to the fuel nozzle than that using the  $C_2$  mechanism. These results are in accord with those reported by Shu et al.<sup>18</sup> for partially premixed flames in a planar configuration. As discussed in Ref. 18, the major effect of  $C_2$  chemistry at lower equivalence ratios,  $\phi = 1.5$ , is to increase the overall reaction rates by increasing the H-atom formation and, thereby, augment the effect of chain-branching reactions by raising the availability of radical species. However, at higher equivalence ratios,  $\phi = 3.0$ , this effect is significantly diminished due to the paucity of O atoms, with the net result that the methane consumption rate is reduced when  $C_2$  mechanism is employed. This is further confirmed by the  $\text{H}_2$  mole fraction profiles shown in Fig. 5. For the  $C_2$  mechanism, the  $\text{H}_2$  mole fraction decreases significantly (note the peak value decreases from 0.04 to 0.017) as  $\phi$  is increased from 1.5 to 3.0. The consumption of  $\text{H}_2$  in the nonpremixed reaction zone occurs through the  $\text{H}_2\text{-O}_2$  chemistry that also produces H and OH radicals. Consequently, a sharp reduction in the  $\text{H}_2$  mole fraction implies a sharp decrease in the radical pool, and this is largely responsible for a much sharper decrease in the methane consumption rate for the  $C_2$  and GRI-2.11 mechanisms compared to that for the  $C_1$  mechanism, as  $\phi$  is increased from 1.5 to 3.0.

The results shown in Figs. 4 and 5 also indicate that the premixed reaction zone obtained using the  $C_2$  mechanism is more sensitive to the equivalence ratio compared with that using the  $C_1$  mechanism. At  $\phi = 1.5$ , this zone for the  $C_2$  mechanism is located closer to the fuel nozzle compared to that for the  $C_1$  mechanism, whereas at  $\phi = 3.0$ , it is located farther from the fuel nozzle. Consequently, at higher level of air premixing,  $\phi = 1.5$ , the  $C_2$  mechanism predicts a larger separation between the two reaction zones compared to the  $C_1$  mechanism. However, at lower level of premixing,  $\phi = 3.0$ , the separation between the reaction zones is larger when the  $C_1$  mechanism is used. In fact, the predictions with the  $C_2$  mechanism indicate that the two reaction zones are beginning to merge at  $\phi = 3.0$ .

Figure 8 presents axial velocity profiles obtained using the  $C_1$  and  $C_2$  mechanisms. For  $\phi = 1.5$ , the lowest axial velocity in the near-exit region of the fuel jet, which can be regarded as the propagation velocity of the premixed flame, is higher for the  $C_2$  mechanism than that for the  $C_1$  mechanism. This is consistent with the results concerning the premixed flame speeds presented in Fig. 2. Note that the conventional premixed flame speed is not meaningful for  $\phi \geq 1.5$  because these equivalence ratios are outside the rich flammability limit. However, the premixed flame speed is meaningful in the context of a partially premixed flame, in which the premixed reaction zone is stabilized due to synergistic interactions between the premixed and nonpremixed reaction zones. These interactions in a partially premixed zone extend the rich flammability limit of the premixed reaction zone.

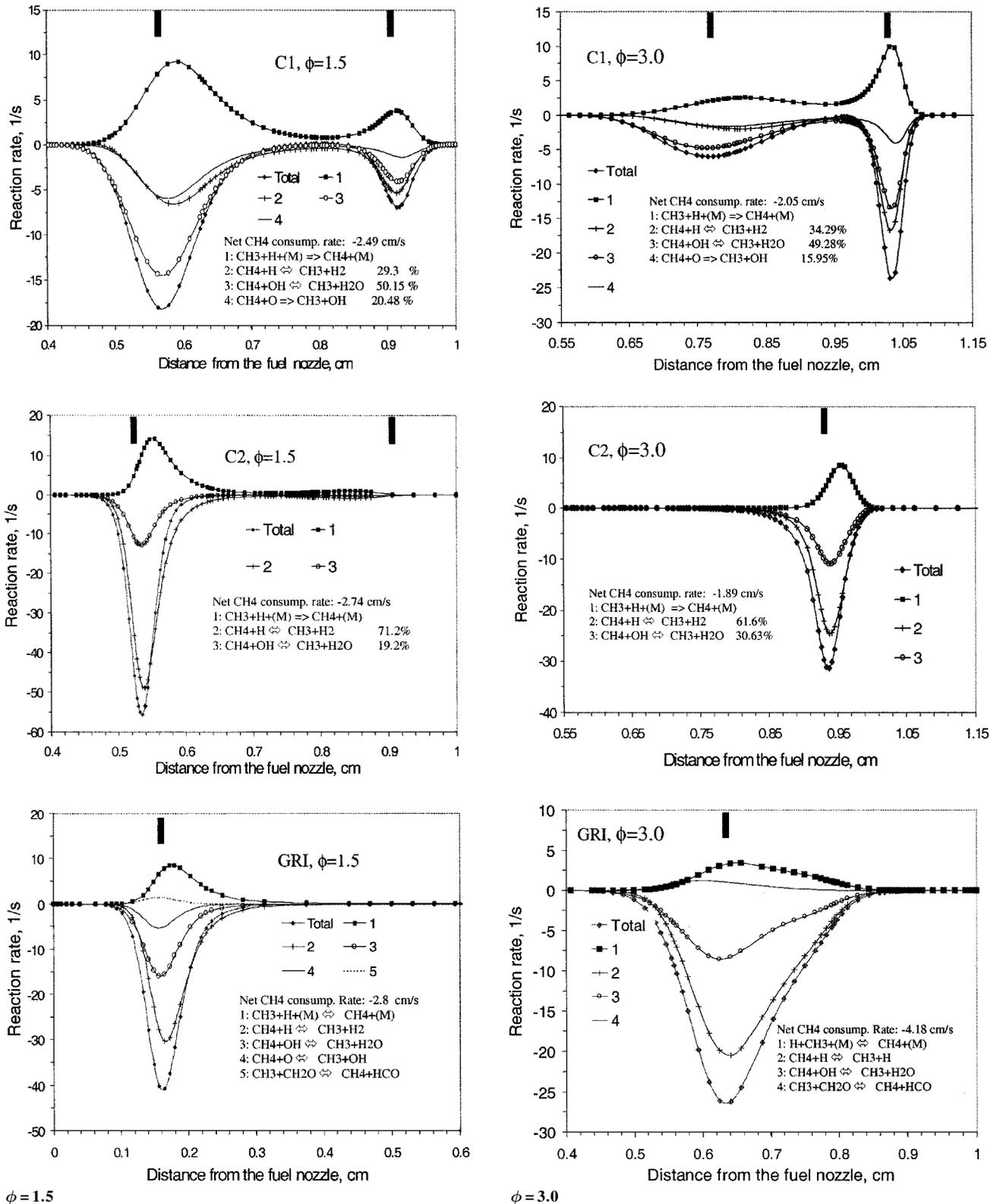


Fig. 9 CH<sub>4</sub> consumption rate profiles for C<sub>1</sub>, C<sub>2</sub>, and GRI-2.11 mechanisms for the conditions of Fig. 4, where, for each case, net consumption rate obtained by integrating the local consumption rate along the axial direction is also provided.

At  $\phi = 3.0$ , the premixed propagation velocity for the C<sub>1</sub> mechanism is higher than that for the C<sub>2</sub> mechanism. In addition, for the C<sub>2</sub> mechanism, the stretch rate experienced by the premixed reaction zone at  $\phi = 1.5$  and  $3.0$  is, respectively, higher and lower compared to that for the C<sub>1</sub> mechanism. The differences between the C<sub>1</sub> and C<sub>2</sub> mechanisms as  $\phi$  is varied from  $1.5$  to  $3.0$  imply that certain reactions in these mechanisms are sensitive to the changes in equivalence ratio. This aspect is discussed in the next section.

#### Comparison of Reaction Rate Profiles

In Fig. 9, we plot the methane consumption rate profiles for the C<sub>1</sub>, C<sub>2</sub>, and GRI-2.11 mechanisms. At a higher level of air premixing,  $\phi = 1.5$ , the total methane consumption rate profile for the C<sub>1</sub> mechanism contains two peaks corresponding to the premixed and nonpremixed reaction zones. In contrast, the profiles for the C<sub>2</sub> and GRI-2.11 mechanisms show only one peak corresponding to the premixed zone. In addition, the methane consumption rate in

the premixed zone for these two mechanisms is significantly higher than that for the  $C_1$  mechanism. Consequently, for both the  $C_2$  and GRI-2.11 mechanisms, methane is completely consumed in the premixed region with very little  $CH_4$  escaping into the nonpremixed region. In contrast, when the  $C_1$  mechanism is employed, a significant amount of methane leaks through the premixed zone, and the amount of leakage increases with  $\phi$ . This is consistent with the  $CH_4$  mass fraction profiles presented in Fig. 5. The higher fuel consumption in the premixed reaction zone for the  $C_2$  and GRI-2.11 mechanisms is attributable to the  $C_2$  chemistry, which provides an additional pathway for methyl consumption through the reaction  $CH_3 + CH_3 \rightleftharpoons C_2H_6$ .

As noted earlier, the premixed reaction zone for the GRI-2.11 mechanism is located very close to the fuel nozzle. The total methane consumption rate profiles for  $\phi = 1.5$  corroborate this observation. For this mechanism, the peak value of the methane consumption rate occurs at a distance of 0.17 cm from the fuel nozzle, whereas those for the  $C_1$  and  $C_2$  mechanisms occur at 0.57 and 0.52 cm, respectively.

At the lower level of air premixing,  $\phi = 3.0$ , the two reaction zones are nearly merged when the  $C_2$  and GRI-2.11 mechanisms are employed, whereas a double-flame structure is still observed when the  $C_1$  mechanism is used. Another difference between the  $C_1$  mechanism and the other two pertains to the dominant methane consumption reaction in the premixed reaction zone. For both the  $C_2$  and GRI-2.11 mechanisms, methane consumption by H atoms dominates that by OH radicals, whereas for the  $C_1$  mechanism, methane consumption by hydroxyl radicals is more important. This is again attributable to the  $C_2$  chemistry that contributes to the higher concentration of H atoms for the  $C_2$  and GRI-2.11 mechanisms compared with that predicted by the  $C_1$  mechanism.

The profiles of total production rates of  $CO_2$  and  $H_2O$  (not presented) indicated that at high level of air premixing,  $\phi = 1.5$ , the production of  $CO_2$  and  $H_2O$  occurs mostly in the premixed zone when the  $C_2$  mechanism is employed, whereas for the  $C_1$  mechanism, the production of  $CO_2$  and  $H_2O$  occurs in both the premixed and nonpremixed zones. This difference is because the methane consumption rate in the premixed reaction zone is much higher for the  $C_2$  mechanism than that for the  $C_1$  mechanism. As a consequence, the heat release rate in the premixed zone is much higher for the  $C_2$  mechanism than that for the  $C_1$  mechanism, as shown in Fig. 10.

As indicated in Figs. 9 and 10, at  $\phi = 3.0$ , the two reaction zones are nearly merged when the  $C_2$  mechanism is employed, whereas a double-flame structure still exists when the  $C_1$  mechanism is used. The implication is that, compared to the  $C_1$  mechanism, the two reaction zones merge at a lower equivalence ratio when either the  $C_2$  or GRI-2.112 mechanism is used. Figures 9 and 10 also show that, for both the  $C_1$  and  $C_2$  mechanisms, the premixed zone chemistry is relatively more dominant at higher levels of air premixing, that is, lower  $\phi$ , whereas the nonpremixed zone chemistry becomes more dominant at lower level of air premixing, that is, higher  $\phi$ .

### Effect of Strain Rate on Flame Structure

Figure 11 shows the flame structure, computed by using the  $C_2$  mechanism, in terms of temperature profiles for three different strain rates at  $\phi = 1.5$ . The strain rate is increased by increasing the exit velocity of both the jets. As stated earlier, the strain rate is defined as the maximum value of the oxidizer-side velocity gradient just before the flame.<sup>9</sup> The strain rates of 21.7, 56.1, and 88.3  $s^{-1}$  were obtained by increasing the jet exit velocities to 20, 40, and 60 cm/s, respectively. As shown in Fig. 11, the separation between the two reaction zones decreases as the strain rate is increased. As noted earlier, the reaction zones also tend to merge as the level of air premixing decreases, that is,  $\phi$  increases from 1.5 to 3.0. There are, however, differences in the change of the partially flame structure caused by increasing the equivalence ratio compared to those caused by increasing the strain rate.

A comparison of Figs. 11 and 4 indicates that the temperature gradient on the fuel side increases as  $\alpha_s$  is increased, whereas it decreases as  $\phi$  is increased. The gradients of major species mole fractions follow a similar behavior. The changes in the gradients of temperature and species mole fractions can be attributed to a balance

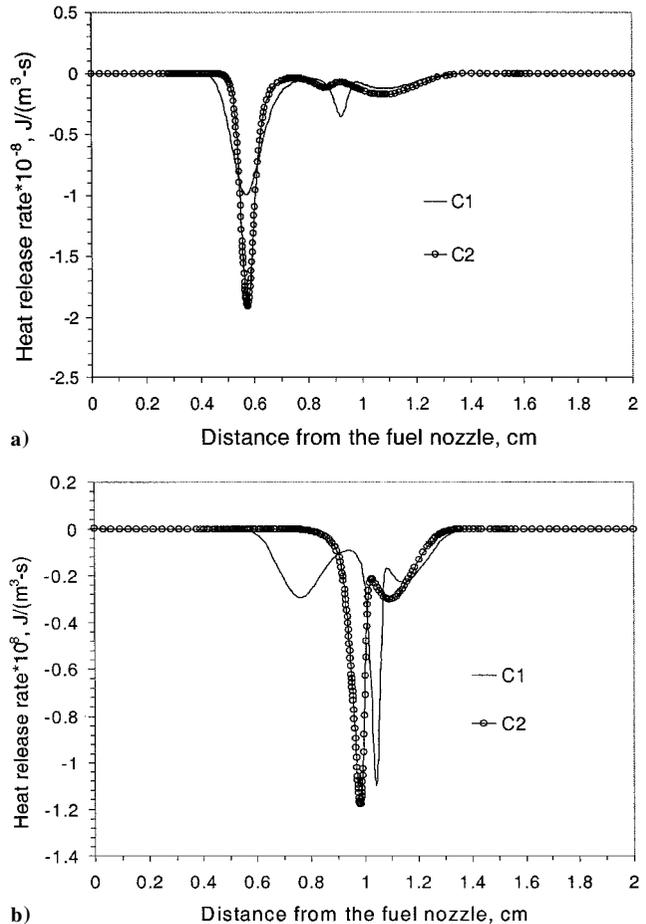


Fig. 10 Heat release rate profiles for the  $C_1$  and  $C_2$  mechanisms for the conditions of Fig. 4 at a)  $\phi = 1.5$  and b)  $\phi = 3.0$ .

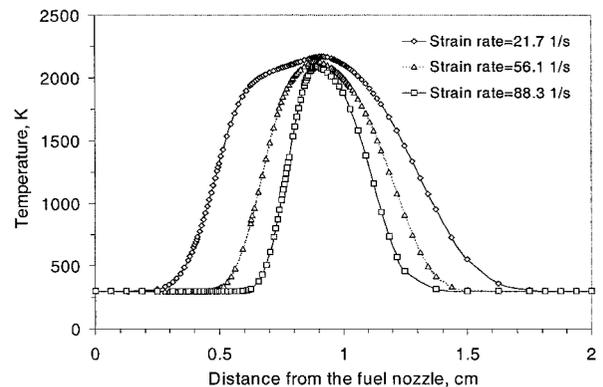


Fig. 11 Temperature profiles predicted by using the  $C_2$  mechanism for three different strain rates at  $\phi = 1.5$ .

between convection, diffusion, and reaction. Increasing the equivalence ratio in the fuel stream directly affects the premixed zone chemistry, which then decreases the physical separation between the premixed and nonpremixed reaction zones. In contrast, increasing the strain rate changes the physical separation between two reaction zones without causing any changes in the reaction chemistry. As the two reaction zones get closer, the heat and radical transport between them is enhanced, which then changes the reaction chemistry of the premixed reaction zone. Thus, the partially premixed flame structure is relatively more sensitive to the equivalence ratio than the strain rate. More specifically, the equivalence ratio affects the structure of the partially premixed flame in the context of changing the premixed flame speed, whereas the strain rate affects reaction chemistry by changing the flame structure aerodynamically.

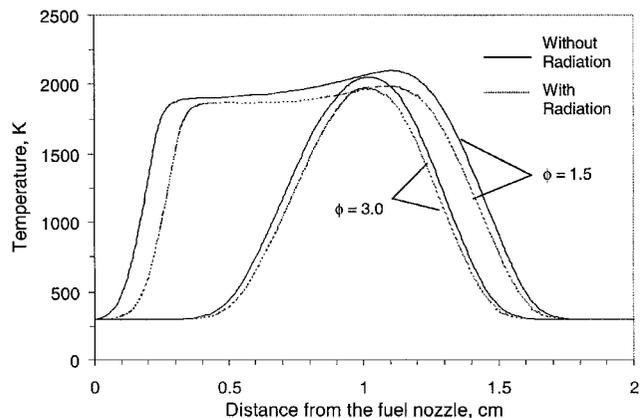


Fig. 12 Temperature profiles for partially premixed flames with and without radiation at  $\phi = 1.5$  and 3.0.

#### Effect of Radiation Heat Transfer on Flame Structure

The effects of thermal radiation on methane–air partially premixed and diffusion flames have been studied by Zhu et al.<sup>24</sup> and Gore et al.<sup>25</sup> Their results showed that radiative heat loss caused by emission from gas molecules changes the temperature and NO mole fraction in low equivalence ratio partially premixed flames significantly. We have employed an optically thin model to examine the effect of radiation heat transfer on the partially premixed flame structure. The Oppdif code was modified to include the volumetric rate of radiation heat loss  $q_r$  in the energy equation. Following Ju et al.,<sup>26</sup>  $q_r$  can be written as

$$q_r = -4\sigma K_p(T^4 - T_\infty^4) \quad (1)$$

$$K_p = P_{\text{CO}_2} K_{\text{CO}_2} + P_{\text{H}_2\text{O}} K_{\text{H}_2\text{O}} + P_{\text{CO}} K_{\text{CO}} + P_{\text{CH}_4} K_{\text{CH}_4} \quad (2)$$

where  $\sigma$  is the Stefan–Boltzmann constant and  $T$  and  $T_\infty$  are the local and ambient temperatures, respectively.  $K_p$  is the Planck mean absorption coefficient of the mixture, and  $P_i$  and  $K_i$  are the partial pressure and Planck mean absorption coefficient of species  $i$ , respectively. The values for  $K_i$  are temperature dependent and are interpolated from Table 1 of Ju et al.<sup>27</sup>

Figure 12 presents a comparison of temperature profiles for flames, computed using the GRI-2.11 mechanism, with and without radiation at  $\phi = 1.5$  and 3.0. The flame structure is not affected significantly, except that the temperature in the two reaction zones is reduced due to the radiation effect. In addition, with the radiation included, the premixed zone moves closer to the nonpremixed zone, although location of the nonpremixed zone is not affected by radiation. A comparison of the major species mole fraction profiles indicated a similar effect of radiation.

#### Effect of Reaction Mechanisms on Flame Extinction

A partially premixed flame will extinguish as the strain rate is increased above a critical value. In Fig. 13, we plot the peak temperature of partially premixed flame vs the strain rate at  $\phi = 1.5$  for the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms. For low to moderate strain rates, the maximum temperature obtained using the GRI-2.11 mechanism is lower than that using the  $C_2$  mechanism, which in turn is lower than that using the  $C_1$  mechanism. The lower values for the GRI-2.11 mechanism are due to the heat loss to the burner because the premixed reaction zone with this mechanism is located very close to the fuel nozzle. The lower maximum temperature for  $C_2$  mechanism compared to that for  $C_1$  mechanism is due to the endothermic reactions associated with  $C_2$  chemistry. At high strain rates, however, the maximum temperature for  $C_1$  mechanism is lower compared to those for the other two mechanisms. This is because the strain rate at extinction is lower for the  $C_1$  mechanism.

The critical strain rates near extinction obtained by using the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms are 450, 878, and 869  $\text{s}^{-1}$ , respectively. The predicted extinction strain rate reported by Tanoff and Smooke<sup>13</sup> for a methane–air partially premixed flames at the same equivalence ratio is about 873  $\text{s}^{-1}$ , which is close to the val-

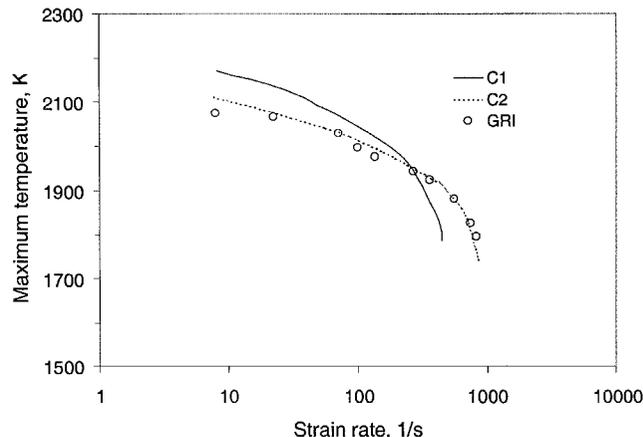


Fig. 13 The maximum flame temperature vs strain rate for the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms at  $\phi = 1.5$ .

ues obtained by using the  $C_2$  and GRI-2.11 mechanisms. Chelliah et al.<sup>9</sup> investigated the extinction of methane–air diffusion flames employing the counterflow flame configuration, and reported an experimental extinction strain rate of 380  $\text{s}^{-1}$ . Their predicted strain rate for extinction for the plug flow boundary conditions, which are also employed in this numerical study, was 391  $\text{s}^{-1}$ . Thus, partially premixed flames are able to sustain significantly higher strain rates than the corresponding diffusion flames. The extinction flame temperature of the partially premixed flame obtained using the  $C_2$  mechanism is 1740 K, which is slightly lower than the predicted value of 1758 K for the diffusion flame.<sup>9</sup>

#### Conclusions

The structure and extinction behavior of counterflow partially premixed flames have been investigated using five different chemistry models. The chemistry models include the  $C_1$  and  $C_2$  mechanisms of Peters and Rogg,<sup>5</sup> the GRI-2.11 mechanism, a 12-step augmented reduced mechanism, and GRI-3.0 mechanism. Simulations have focused on the comparison of these mechanisms in predicting the structure and extinction behavior of methane–air partially premixed flames over a wide range of strain rates and equivalence ratios, including those corresponding to premixed and diffusion flames.

Premixed flame speeds calculated using the  $C_2$  and GRI-2.11 mechanisms are in good agreement with the experimental data, whereas those obtained using the  $C_1$  mechanism show significant differences at fuel-rich conditions. However, the flammability limits ( $0.5 < \phi < 1.4$ ) for methane–air mixtures obtained using these three mechanisms are almost identical. In addition, the diffusion flame structures computed using these mechanisms are essentially the same, except for small differences in the peak temperature values.

Results pertaining to partially premixed flames indicate all five mechanisms reproduce qualitatively the double-flame structure associated with these flames. There are, however, significant quantitative differences in the flame structures obtained using the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms. For low to moderate strain rates and high to moderate levels of air premixing, the GRI-2.11 mechanism yields a flame structure that is markedly different from that obtained using the other two mechanisms. For example, for a partially premixed flame established at  $\phi = 1.5$  and a strain rate of 7.5  $\text{s}^{-1}$ , the rich premixed reaction zone for the GRI-2.11 mechanism is located very close to the fuel nozzle, which causes significant heat loss to the nozzle. In addition, the global flame for this mechanism is significantly broader compared to those for the other two mechanisms. This is somewhat unexpected but noteworthy result, especially because the predictions of  $C_2$  and GRI-2.11 mechanisms are in good agreement for both the premixed and diffusion flames. Experimental data are needed to resolve these differences.

There are important quantitative differences in the structures of partially premixed flames simulated using the  $C_1$  and  $C_2$  mechanisms. First, the methane consumption rate in the premixed reaction zone is higher for the  $C_2$  mechanism compared to that for the  $C_1$  mechanism. Consequently, the formation rates of  $\text{CO}_2$  and  $\text{H}_2\text{O}$ , as

well as the heat release rate in the premixed zone, are also higher for the  $C_2$  mechanism. This is directly attributable to the  $C_2$  chemistry, which provides an additional pathway for methyl consumption through the reaction  $CH_3 + CH_3 \rightleftharpoons C_2H_6$ . Second, the partially premixed flame structure obtained using the  $C_2$ -mechanism exhibits stronger sensitivity to the equivalence ratio compared with that using the  $C_1$  mechanism.

As the equivalence ratio  $\phi$  and/or the strain rate  $a_s$  are increased, the two reaction zones move closer and eventually merge into a single zone. The predicted values of  $\phi$ , at which the merging occurs, is smaller for the  $C_2$  and GRI-2.11 mechanisms compared to that for the  $C_1$  mechanism. For a given  $a_s$ , a further increase in  $\phi$  yields a diffusion flame structure. On the other hand, a further increase in  $a_s$  (for a given  $\phi$ ) leads to flame extinction. The extinction strain rates obtained using the  $C_2$  and GRI-2.11 mechanisms are significantly higher than that using the  $C_1$  mechanism. For example, at  $\phi = 1.5$ , the extinction strain rates are 450, 878, and  $869 \text{ s}^{-1}$  for the  $C_1$ ,  $C_2$ , and GRI-2.11 mechanisms, respectively.

An optically thin model has been used to examine the effect of radiation on the partially premixed flame structure. Results indicate that the flame structure is not affected significantly due to radiation, except that the temperature in the two reaction zones is reduced, and that the premixed zone is located slightly closer to the nonpremixed zone when the radiation is included.

Finally, the effect of an equivalent laminar flame speed on partially premixed flame structure has been examined. It is observed that the rich premixed zone has a propagation velocity that can be characterized in terms of an equivalent laminar flame speed corresponding to a given equivalence ratio. Because the velocity gradient on the fuel side is determined by the combined effects of this equivalent flame speed and the fuel jet exit velocity, the premixed flame speed plays an important role in determining the stretch rate and, therefore, the structure of partially premixed flames.

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